

## CRITICAL ITEMS LIST

REFERENCE DESIGNATOR: HPA-2  
 NAME / QUANTITY: Jettison Handle  
 DRAWING REFERENCE: 417969

PROJECT: HST  
 LRU NAME / QUANTITY: HST PFR/APC Assembly  
 LRU PART NUMBER: 6ED3A19295-001#00

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 SUBSYSTEM: N/A  
 EFFECTIVITY: ALL ORBITERS

FAILURE MODE NUMBER HST-HPA-2-1	CRITICALITY 1R/2	FAILURE EFFECT	RETENTION RATIONALE
<b>FUNCTION</b>  Jettison Handle assist the crew with the contingency jettison of the Solar Arrays		<b>END ITEM</b> Jettison Handle is free in the payload bay.	I. Design Feature to Minimize the Chance of the Failure Mode  A. Design The Jettison Handle was designed to an ultimate structural safety factor of 1.4.
<b>FAILURE MODE AND CAUSE</b>  <b>MODE</b> Jettison Handle comes loose in the payload bay.  <b>CAUSE(S)</b> 1) Pip Pin Failure 2) Vibration		<b>MISSION</b> None.	B. Tolerances Sufficient tolerances were used in the PFR Extender design to prevent jamming by expansion and contraction of material due to temperature extremes or on-orbit use.  C. Materials - Major Components  Probe - 15-SPH, Condition H1025, Pip pin - Modified MS stainless steel pin (PIN 4173211)
<b>REDUNDANCY SCREENS</b> A - Pass B - Pass C - Pass	<b>REMAINING PATHS</b> 1.) Hitch Pin	<b>CREW / VEHICLE</b> Loss of crew and vehicle due to damaged created from loose Jettison Handle in the payload bay.	II. Testing and Analysis  A. Acceptance Testing  1. PIA  A full pre-installation acceptance (PIA) test was performed on the Jettison Handle assembly before it was delivered to KSC to support flight. The PIA verified that the Jettison Handle is functioning within tolerances and that the assembly is clean (ref. 189320299).  2. Pip Pin Acceptance  The Pip pin was used in the STS-31 manned thermal vacuum test to demonstrate its operation under thermal conditions. The operation was successful at -90°F.
<b>MISSION PHASE</b>	<b>CORRECTIVE ACTION TIMES</b>		
Launch/Landing	Minutes	Seconds	

## CRITICAL ITEMS LIST

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SUBSYSTEM: N/A

EFFECTIVITY: ALL ORBITERS

REFERENCE DESIGNATOR: HPA-1  
 NAME/QUANTITY: Jettison Handle  
 DRAWING REFERENCE: 417962

PROJECT: HST  
 LRU NAME/QUANTITY: HST PFR/APC Assembly  
 LRU PART NUMBER: 96D 381962-Sub503

FAILURE MODE NUMBER HST-HPA-2-1	CRITICALITY 1R/2	FAILURE EFFECT	RETENTION RATIONALE
<b>FUNCTION</b>  Jettison Handle assist the crew with the contingency jettison of the Solar Arrays.		<b>END ITEM</b> Jettison Handle is free in the payload bay.	<b>6. Certification Testing</b>  1. Thermal Vacuum  The Jettison Handle was exposed to a cold temperature (-132°F) vacuum ( $1 \times 10^{-5}$ torr) environment. This test was used to check the tolerances of the hex probe to the PFR socket and the operation of the pip pin. The operational requirement was -90°F (Ref. JSC-23550)
<b>FAILURE MODE AND CAUSE</b> <b>MODE</b> Jettison Handle comes loose in the payload bay.  <b>CAUSE(S)</b> 1) Pip Pin Failure 2) Vibration		<b>MISSION</b> None.	<b>2. Functionals</b>  The Jettison Handle pip pin was functionally operated prior to and immediately after all acceptance/certification tests to verify that the test environment did not degrade the hardware performance.
<b>REduNDANCY SCREENS</b>	<b>REMAINING PATHS</b>	<b>CREW / VEHICLE</b> Loss of crew and vehicle due to damage created from loose Jettison Handle in the payload bay.	
A - Pass B - Pass C - Pass	1.) Hitch Pin	<b>INTERFACE</b> EVA-plated PFR socket.	
<b>MISSION PHASE</b>	<b>CORRECTIVE ACTION TIMES</b>		
	<b>TIME TO EFFECT</b>	<b>TIME TO CORRECT</b>	
Launch/Landing	Minutes	Seconds	

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## CRITICAL ITEMS LIST

REFERENCE DESIGNATOR: HPA-2  
 NAME / QUANTITY: Jettison Handle  
 DRAWING REFERENCE: 410060

PROJECT: HST  
 LRU NAME / QUANTITY: HST PFR/APC Assembly  
 LRU PART NUMBER: SED3019886-601/98

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 SUBSYSTEM: MA  
 EFFECTIVITY: ALL ORBITERS

FAILURE MODE NUMBER	CRITICALITY	FAILURE EFFECT	RETENTION RATIONALE											
HST-HPA-2-1	1R/2													
<b>FUNCTION</b>		<b>END ITEM</b> Jettison Handle is free in the payload bay.	<b>C. Certification Analysis</b> All Jettison Handle components were analyzed to the following induced environments to verify that the assembly can withstand the environment levels: <table> <tr> <td><b>1. Requirements</b></td> <td><b>Source</b></td> <td><b>Data</b></td> </tr> <tr> <td>A. <b>Structures</b></td> <td>F29-7064 (stress analysis) JSC-25838</td> <td>300 lbs all directions OK per Matl-93-079</td> </tr> <tr> <td>B. <b>Temperature</b></td> <td>LESC-30943 JSC-29550</td> <td>+250°F -90°F</td> </tr> </table>			<b>1. Requirements</b>	<b>Source</b>	<b>Data</b>	A. <b>Structures</b>	F29-7064 (stress analysis) JSC-25838	300 lbs all directions OK per Matl-93-079	B. <b>Temperature</b>	LESC-30943 JSC-29550	+250°F -90°F
<b>1. Requirements</b>	<b>Source</b>	<b>Data</b>												
A. <b>Structures</b>	F29-7064 (stress analysis) JSC-25838	300 lbs all directions OK per Matl-93-079												
B. <b>Temperature</b>	LESC-30943 JSC-29550	+250°F -90°F												
<b>FAILURE MODE AND CAUSE</b>		<b>MISSION</b> None.												
<b>MODE</b> Jettison Handle comes loose in the payload bay.		<b>CREW / VEHICLE</b> Loss of crew and vehicle due to damaged created from loose Jettison Handle in the payload bay.												
<b>CAUSES(S)</b> 1) Pin Pin Failure 2) Vibration		<b>INTERFACE</b> EVA-plate PFR socket.												
<b>REDUNDANCY SCREENS</b>	<b>REMAINING PATHS</b>													
A - Pass B - Pass C - Pass	1.) Hitch Pin													
<b>MISSION PHASE</b>	<b>CORRECTIVE ACTION TIMES</b>													
	<b>TIME TO EFFECT</b>	<b>TIME TO CORRECT</b>												
Launch/Landing	Minutes	Seconds												

## CRITICAL ITEMS LIST

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SUBSYSTEM: MA

EFFECTIVITY: ALL ORBITERS

REFERENCE DESIGNATOR: HPA-2  
 NAME / QUANTITY: Jettison Handle  
 DRAWING REFERENCE: 417060

PROJECT: HST  
 LRU NAME / QUANTITY: HST PFR/APC Assembly  
 LRU PART NUMBER: 9ED 391 9026-001000

FAILURE MODE NUMBER	CRITICALITY	FAILURE EFFECT	RETENTION RATIONALE
HST-HPA-2-1	1R/2		
<b>FUNCTION</b>  Jettison Handle assist the crew with the contingency jettison of the Solar Arrays		<b>END ITEM</b>  Jettison Handle is free in the payload bay.	<b>III. Inspection</b>  <b>A. Manufacturing</b>  1. The Jettison Handle components were inspected prior to build-up for conformance to their applicable drawings.  <b>B. Assembly</b>  1. Jettison Handle and pin are cleaned and Inspected to the levels described in JSC 5322B. Once cleaned, the Jettison Handle was bagged to prevent anything from contaminating the unit.  <b>C. Testing</b>  1. The hardware was fully inspected for any signs of galling as a part of the pre/post functional tests performed prior to and immediately after all certification and acceptance tests.
<b>FAILURE MODE AND CAUSE</b>  <b>MODE</b> Jettison Handle comes loose in the payload bay.  <b>CAUSE(S)</b> 1) Pin Failure 2) Vibration		<b>MISSION</b> None.	
<b>REDUNDANCY SCREENS</b> A - Pass B - Pass C - Pass	<b>REMANANT PATHS</b> 1.) Hitch Pin	<b>CREW / VEHICLE</b> Loss of crew and vehicle due to damage created from loose Jettison Handle in the payload bay.	
<b>MISSION PHASE</b>	<b>CORRECTIVE ACTION TIMES</b>	<b>INTERFACE</b> EVA-plate PFR socket	
Launch/Landing	Minutes	Seconds	

## CRITICAL ITEMS LIST

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 SUBSYSTEM: N/A  
 EFFECTIVITY: ALL ORBITERS

REFERENCE DESIGNATOR: HPA-2  
 NAME / QUANTITY: Jetison Handle  
 DRAWING REFERENCE: 4178460

PROJECT: HST  
 LRU NAME / QUANTITY: HST PFR/APC Assembly  
 LRU PART NUMBER: 96034118294-001000

FAILURE MODE NUMBER	CRITICALITY	FAILURE EFFECT	RETENTION RATIONALE
HST-HPA-2-1	1R/2		
<b>FUNCTION</b>  Jetison Handle assist the crew with the contingency jettison of the Solar Arrays.		<b>END ITEM</b> Jetison Handle is free in the payload bay.	<b>IV. Failure History</b> A. None, HST PFR/APC flew on STS-31, but was not used.  <b>V. Operations</b> A. <b>Effects of Failure</b> Jetison Handle loses connection to the APC and is free to move within the payload bay. B. <b>Crew Actions</b> None. C. <b>Training</b> None. D. <b>Mission Constraints</b> Possible damage to cargo within payload bay if the Jetison Handle does come loose. E. <b>In-Flight Check-Outs</b> None
<b>FAILURE MODE AND CAUSE</b>  <b>MODE</b> Jetison Handle comes loose in the payload bay.  <b>CAUSE(S)</b> 1) Pin Pin Failure 2) Vibration		<b>MISSION</b> None.  <b>CREW / VEHICLE</b> Loss of crew and vehicle due to damage created from loose Jetison Handle in the payload bay.	
<b>REDUNDANCY SCREENS</b> A - Pass B - Pass C - Pass	<b>REMAINING PATHS</b> 1.) Hitch Pin	<b>INTERFACE</b> EVA-plate PFR socket.	
<b>MISSION PHASE</b>	<b>CORRECTIVE ACTION TIMES</b>		
	<b>TIME TO EFFECT</b>	<b>TIME TO CORRECT</b>	
Launch/Landing	Minutes	Seconds	

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